

Name:	
Enrolment No:	

UNIVERSITY OF PETROLEUM AND ENERGY STUDIES

End Semester Examination, December 2018

Programme Name: B. Tech Mechanical

Semester : III

Course Name : Material Science

Time : 03 hrs

Course Code : MEMA 2001

Max. Marks: 100

Nos. of page(s) : 02

**Instructions: 1. Assume data only if required and mention the assumption clearly.
2. Use graph sheet for Q8.**

SECTION A : 20 marks

S. No.		Marks	CO
Q 1	Sketch completely labelled ideal & real cooling curves for pure substance/metal.	4	CO2
Q 2	Define superalloys and ceramics.	4	CO4
Q 3	Discuss various fatigue cycles.	4	CO4
Q 4	a. Differentiate between ferrous & non-ferrous alloys. b. Define creep.	4	CO4
Q 5	Differentiate between toughness & resilience.	4	CO1

SECTION B : 40 marks

Q 6	Define composites. Classify composites and give one example of each.	10	CO4																												
Q 7	Discuss Griffith's theory of fracture. Derive the expression for critical stress for crack propagation.	10	CO4																												
Q 8	Data for tensile test for an alloy is given below. Using this data: a. Plot labelled engineering stress vs strain curve (mark yield point, UTS, elastic region, plastic region, fracture point). b. Calculate Young's Modulus of the alloy.	6+4	CO1																												
	<table border="1" style="width: 100%; border-collapse: collapse;"> <thead> <tr> <th style="width: 25%;">Engineernig Stress (MPa)</th> <th style="width: 25%;">Engineernig Strain</th> <th style="width: 25%;">Engineernig Stress (MPa)</th> <th style="width: 25%;">Engineernig Strain</th> </tr> </thead> <tbody> <tr> <td>0</td> <td>0</td> <td>524</td> <td>0.08</td> </tr> <tr> <td>210</td> <td>0.001</td> <td>515</td> <td>0.10</td> </tr> <tr> <td>380</td> <td>0.002</td> <td>500</td> <td>0.12</td> </tr> <tr> <td>415</td> <td>0.005</td> <td>475</td> <td>0.14</td> </tr> <tr> <td>470</td> <td>0.01</td> <td>448</td> <td>0.16</td> </tr> <tr> <td>490</td> <td>0.02</td> <td>385</td> <td>0.18</td> </tr> </tbody> </table>	Engineernig Stress (MPa)	Engineernig Strain	Engineernig Stress (MPa)	Engineernig Strain	0	0	524	0.08	210	0.001	515	0.10	380	0.002	500	0.12	415	0.005	475	0.14	470	0.01	448	0.16	490	0.02	385	0.18		
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	510 517	0.04 0.06	355	0.19 (Fracture)		
Q 9	<p>Calculate the density of fcc platinum in g/cm^3 if its lattice constant 'a' is 0.39239 nm and atomic mass is 195.09 g/mol.</p> <p style="text-align: center;">OR</p> <p>A sample of bcc metal was placed in an x-ray diffractometer and x-rays of wavelength 0.1541 nm were used. Diffraction from (221) planes was obtained at $2\theta = 88.838^\circ$. Calculate value of lattice constant 'a' for this bcc metal. Assume first order diffraction.</p>				10	CO1
SECTION C: 40 marks						
Q 10	<p>A. Sketch completely labelled TTT curve.</p> <p>B. Under what cooling conditions Martensite forms?</p> <p>C. How crystal structure of Martensite is formed on cooling?</p> <p>D. What type of pearlite grains are formed during annealing and normalizing processes? Why?</p> <p style="text-align: center;">OR</p> <p>E. Define Hardenability.</p> <p>F. Discuss Jominy End Quench test.</p> <p>G. Discuss Martempering, Austempering and Nitriding Processes.</p>				10 2 4 4 4 4 12	CO3
Q 11	<p>Given below is the phase diagram of Pb-Sn alloy. If one Kg of 70 % Pb- 30 % Sn alloy is cooled from 300 °C , calculate:</p> <p>a. The weight percent of liquid and alpha at 250 °C.</p> <p>b. The weight percent and weight in kilograms of liquid and alpha just above the eutectic temperature (183 °C).</p> <p>c. The weight percent of alpha and beta at eutectic point.</p>				20	CO2

