

Name:
Enrolment No:



UNIVERSITY OF PETROLEUM AND ENERGY STUDIES
End Semester Examination, May 2019

Programme Name: B.Tech-ADE& Mechanical
Course Name : Computational Fluid Dynamics
Course Code : GNEG-403
Nos. of page(s) : 03
Instructions : Attempt all questions.Q8 and Q11 are having internal choice.

Semester : VIII
Time : 03 hrs
Max. Marks : 100

SECTION A(20 Marks)

S. No.		Marks	CO
Q1	What is the physical meaning of divergence of velocity? State it clearly.	4	CO1
Q2	Show that second order Wave equation is a hyperbolic in nature. Also, signify the meaning of being hyperbolic in nature. $\frac{\partial^2 u}{\partial t^2} = c^2 \frac{\partial^2 u}{\partial x^2}$	4	CO2
Q3	Using polynomial approach drive the finite difference expression for the derivative at the boundary (one sided difference).	4	CO3
Q4	Assess the Central Differencing Scheme on the following properties Conservativeness, Boundedness, Transportiveness and Accuracy.	4	CO4
Q5	Consider a one dimensional steady state heat conduction equation without external source. Apply the Galerkin weak formulation and obtain second order accurate for linear element.	4	CO5

SECTION B(40 Marks)

Q6	Deduce Energy equation in the conservation form.	10	CO1
Q7	Consider a function $\phi(x, y) = e^x - y^2$. Consider the point $(x,y)=(1,1)$. a) Calculate exact value of $\frac{\partial \phi}{\partial x}$ and $\frac{\partial \phi}{\partial y}$ at this point. b) Tabulate percentage difference in the result of $\frac{\partial \phi}{\partial x}$ and $\frac{\partial \phi}{\partial y}$ with respect to exact value, when solved using first order forward difference, first order backward difference and second order central difference. Take $\Delta x = \Delta y = 0.01$	10	CO3
Q8	Comprehend $k - \epsilon$ and $k - \omega$ turbulence model in details. OR Comprehend Large eddy simulation for turbulence modeling in detail.	10	CO4
Q9	Compare SIMPLE and SIMPLER algorithm with the governing equations. Also, draw flow chart for both of the algorithm.	10	CO5

SECTION-C

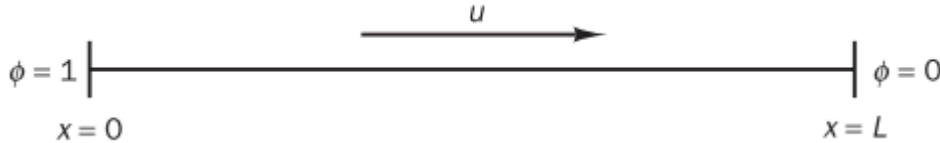
Q10 A property ϕ is transported by means of convection and diffusion through the one-dimensional domain sketched below. Using five equally spaced cells calculate distribution of ϕ as a function of x while using QUICK scheme for the following cases.
 Case 1- $u=0.1$ m/s.
 Case 2- 2.5 m/sec.
 Compare the results obtained with analytical solution. The following data apply $L= 1$ m, $\rho= 1$ kg/m³, $\tau=0.1$ kg/m. s.

Take Analytical formulation $\frac{\phi - \phi_0}{\phi_L - \phi_0} = \frac{\exp\left(\frac{\rho u x}{\tau}\right) - 1}{\exp\left(\frac{\rho u L}{\tau}\right) - 1}$

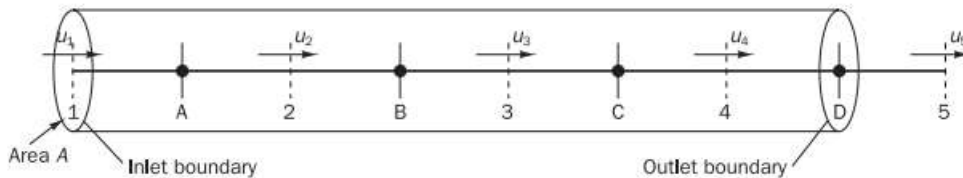
Take Governing Equation $\frac{d}{dx}(\rho u \phi) = \frac{d}{dx}\left(\tau \frac{d\phi}{dx}\right)$

20

CO4



Q11 Consider steady, one-dimensional flow of a constant density fluid through a circular duct with constant cross section. Take the staggered grid as shown in figure given below. Use SIMPLE algorithm and predict pressure at different pressure nodes. Also, obtain corrected velocity field at velocity nodes.
Use following data
Density=1 kg/m³. Duct area Constant, d=1, boundary conditions $u_1= 10$ m/s, $p_d= 0$ pa, initial guessed velocity $u_2^*=8$ m/s, $u_3^*=11$ m/s and $u_4^*=7$ m/s. Compare the computed result against the exact solution of $u_2=u_3=u_4=10$ m/sec.



20

CO5

OR

A planar two-dimensional nozzle is shown in figure given below. The flow is steady and frictionless and the density of the fluid is constant. Use the backward staggered grid with five pressure nodes and four-velocity node. The stagnation pressure is given at the inlet and the static pressure is specified at the exit. Using, SIMPLE algorithm, write down the discretized momentum and pressure correction equations and solve the unknown pressure and velocities at the nodes. Check whether the computed velocity field satisfy continuity and evaluate the error in computed pressure and velocity field by comparing with the exact solution.

Use following data

**Density=1 kg/m³. Duct area Constant, L=2 m,
P₀=10 Pa, p_E=0 Pa. mass flow rate= 1 kg/sec.**

